

China Lunar Exploration Program

Global Space Development Summit
November, 2009

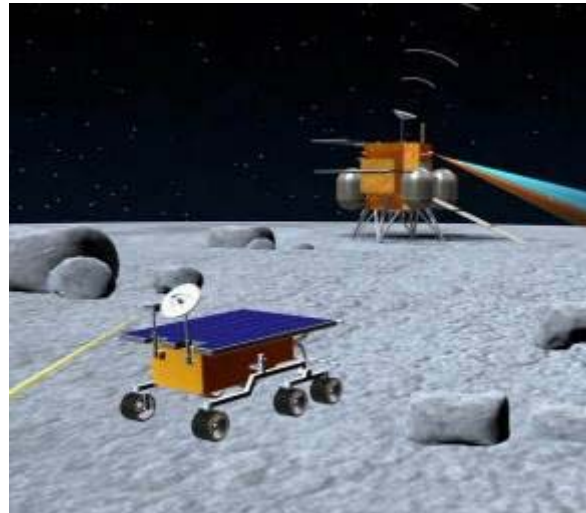


Plan of CLEP

Three phases in China Lunar Exploration Program (CLEP), or Chang'E program: orbiting, landing and sample returning.



Orbiting



Landing



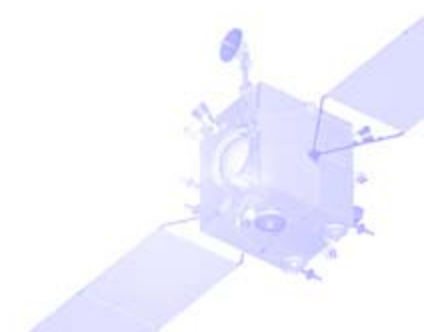
sample returning



1. 1st Phase of CLEP

Engineering Goals of 1st phase of CLEP is to develop a lunar orbiter and perform lunar remote sensing:

- ① To develop the first Chinese lunar orbiter
- ② To develop technology for lunar exploration
- ③ To develop research of lunar science
- ④ To build lunar exploration engineering system
- ⑤ To accumulate experience for further exploration





Scientific goals of 1st phase of CLEP

- ① To acquire 3 dimensional image of lunar surface;
- ② To analyze abundance and distribution of different elements on lunar surface;
- ③ To explore characteristics of lunar soil;
- ④ To explore the environment between the Earth and Moon.





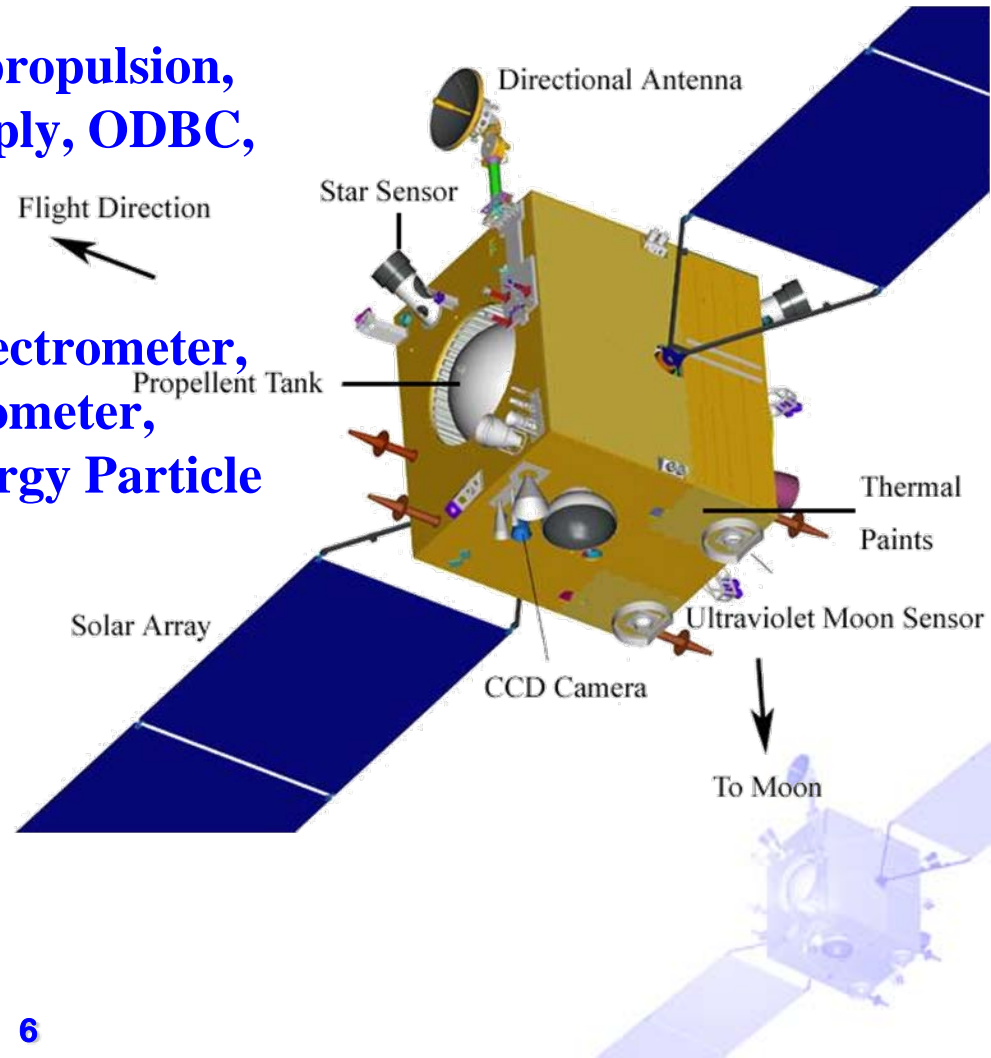
Engineering System:

- Chang'E-1 orbiter
- CZ-3A Launch vehicle
- Xi'chang launch center
- ground telecommunication
- ground application



CE-1 Orbiter

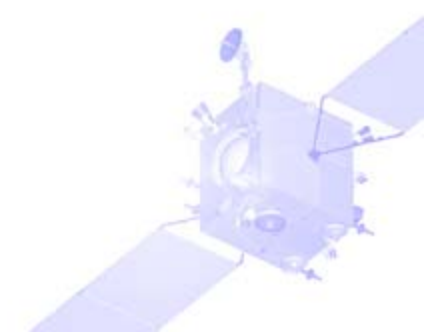
- ⇒ Based on DFH-3 bus
- ⇒ Nine subsystems: structure, GNC, propulsion, TT&C, thermal control, Power supply, ODBC, directional antenna, payloads.
- ⇒ Life: 1 year
- ⇒ Payloads: CCD camera, Imager Spectrometer, Laser altimeter, γ -ray/X-ray spectrometer, Microwave detector, Solar high energy Particle detector, Low energy ion detector





Events of CE-1

1. Oct. 24, 2007, successfully launched
2. Nov. 7, 2007, enter 200km normal operation orbit
3. Dec. 6, 2008, after one year around the Moon, went into 100km lunar orbit
4. Dec. 19, 2008, lower orbit into 15km×100km orbit
5. Mar. 1, 2009, impact lunar surface in 52.36°E , 1.5°S by command, fulfilled 12 times coverage to the moon.



⇒ **Scientific achievements:**
acquire 1st image of lunar
surface by China

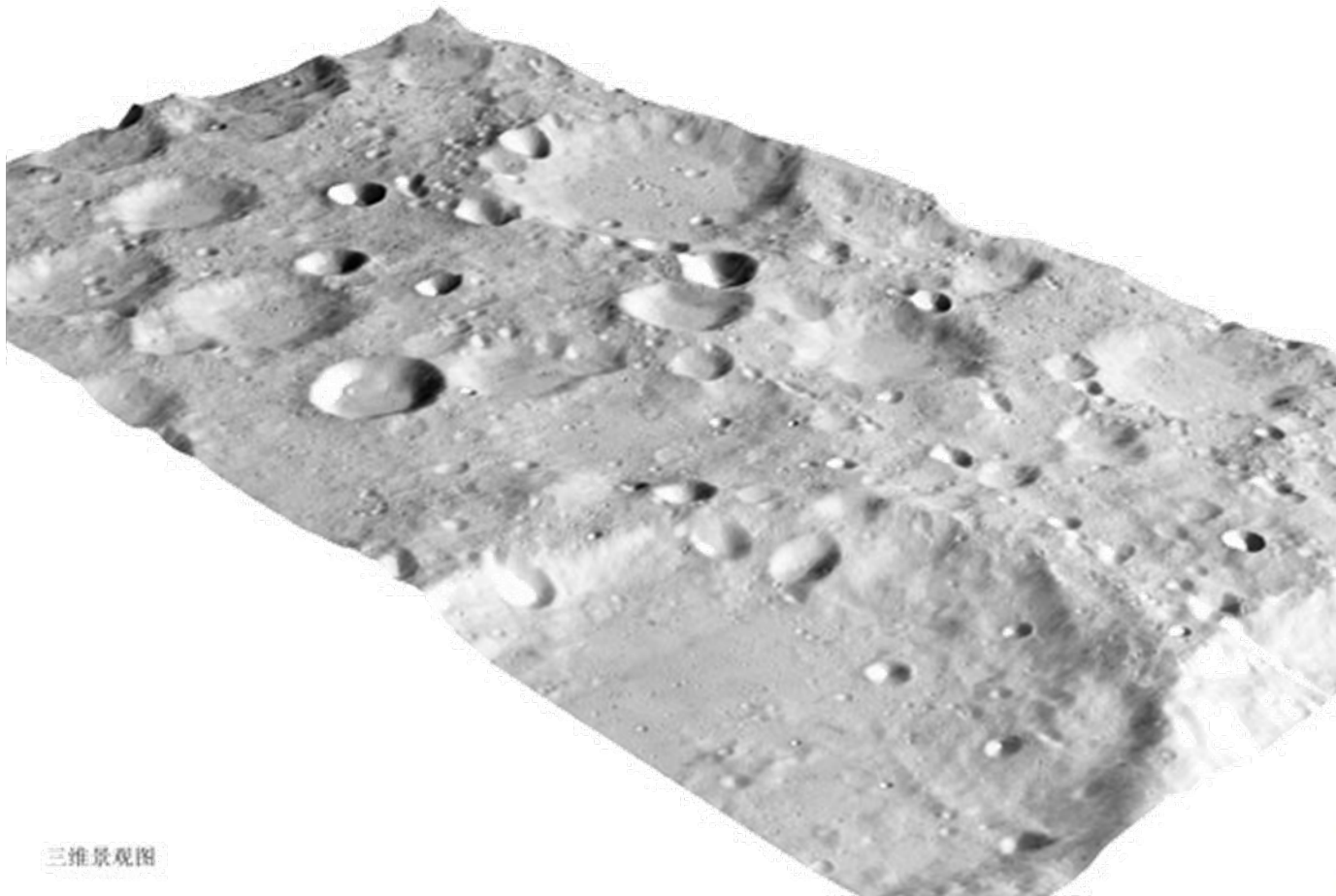
54°S ~ 70°S





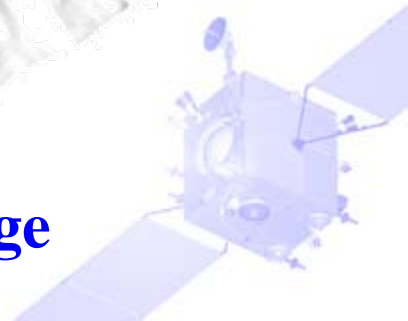
Achievement of CE-1

Acquire high quality 3-D image of lunar globe.



三推景观图

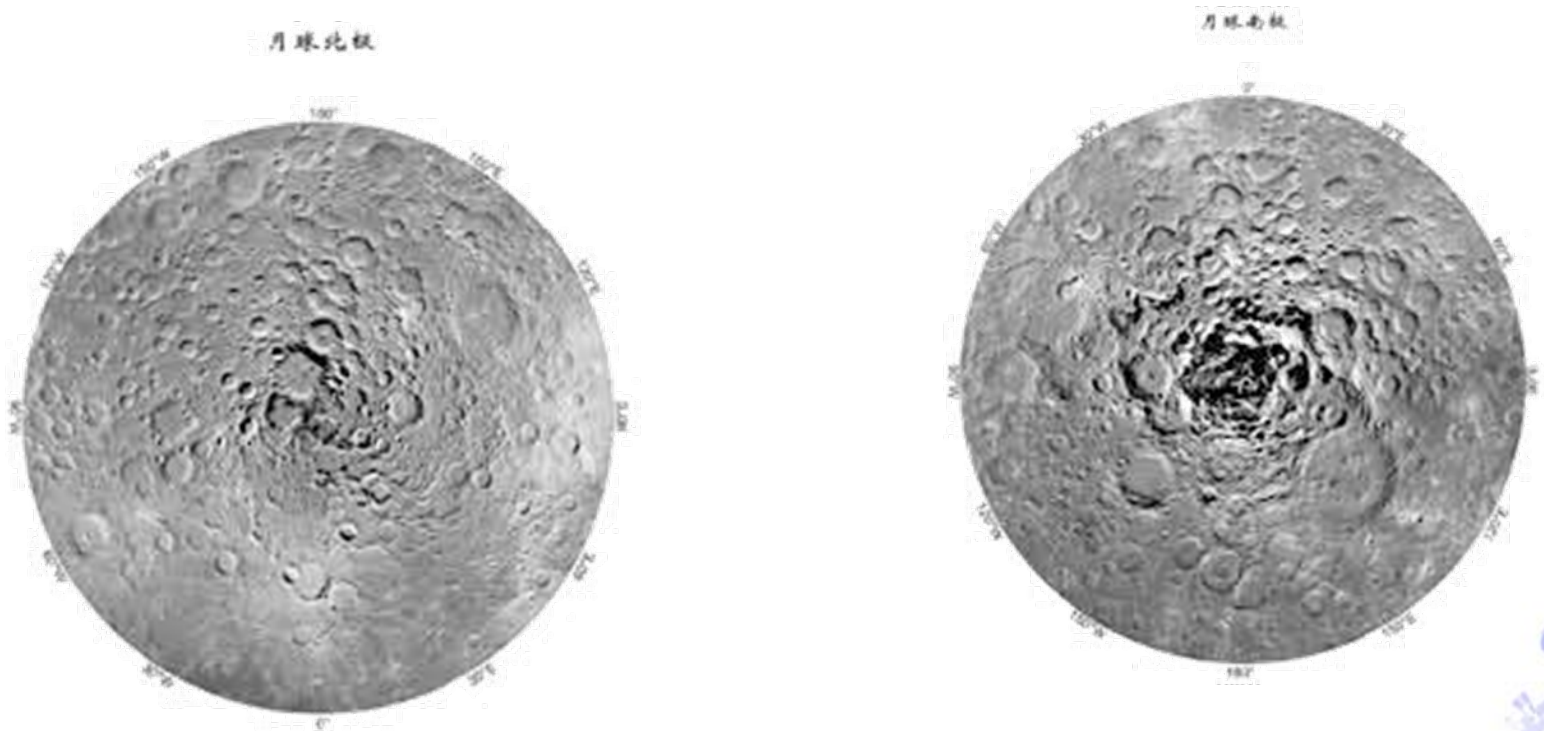
3-D composition of first lunar surface image





Achievement of CE-1

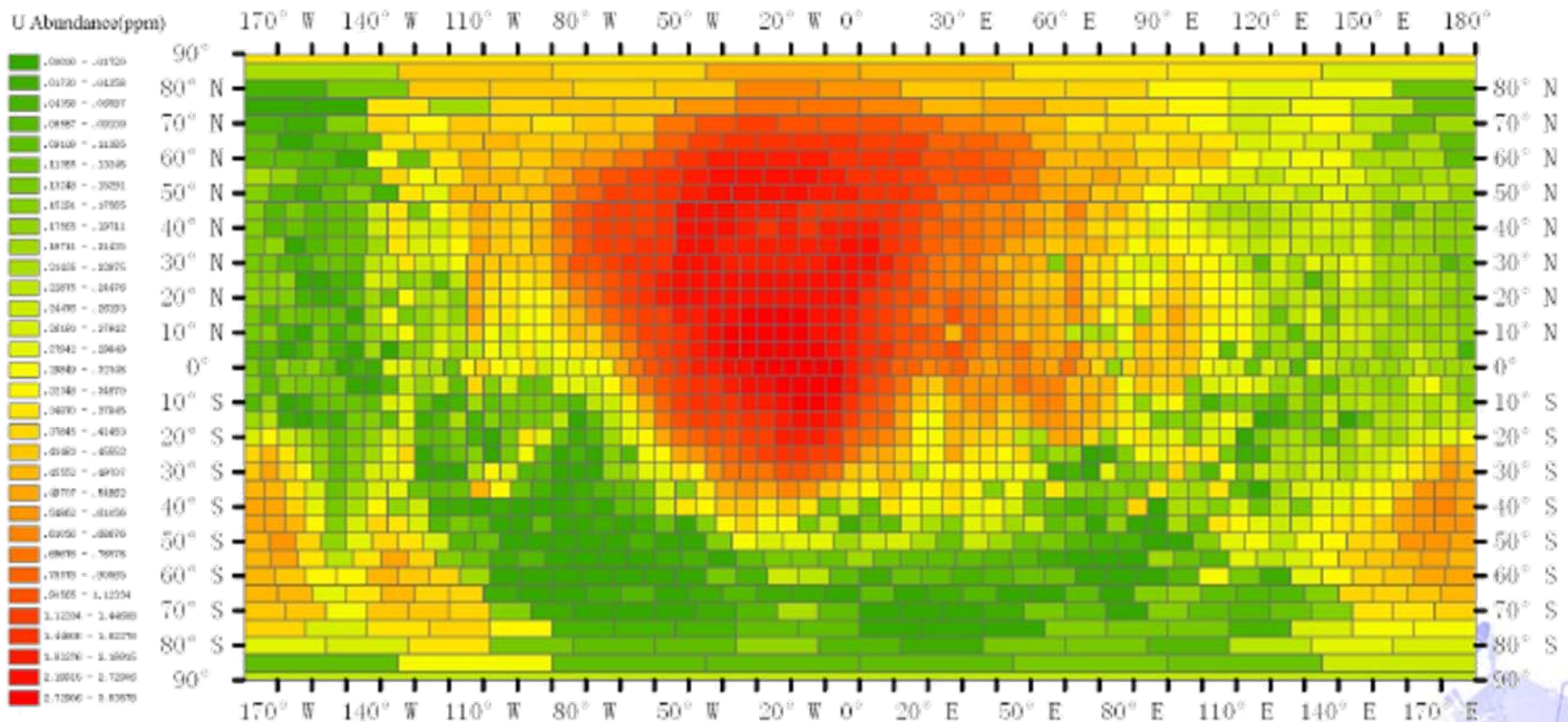
Image of lunar north and south pole.





Achievement of CE-1

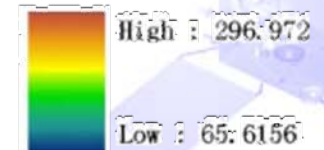
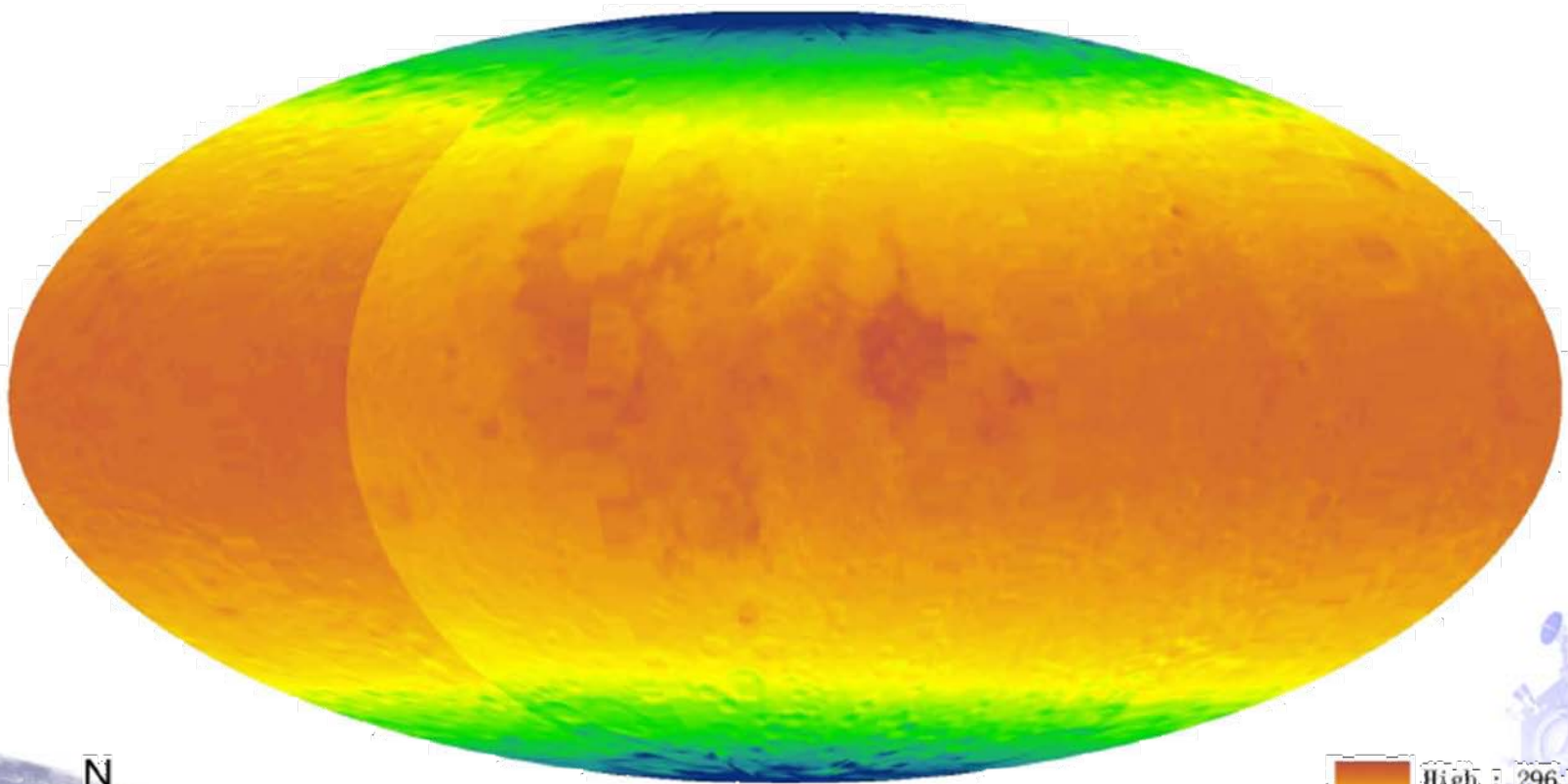
Distribution of elements of lunar globe.





Achievement of CE-1

Brightness Temperature of lunar surface

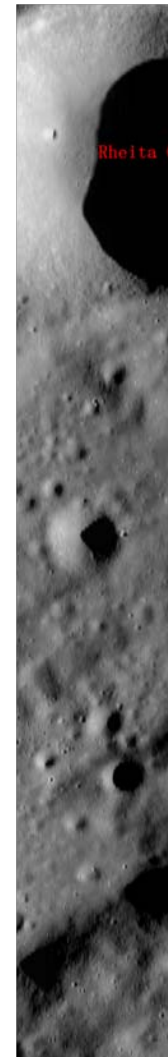
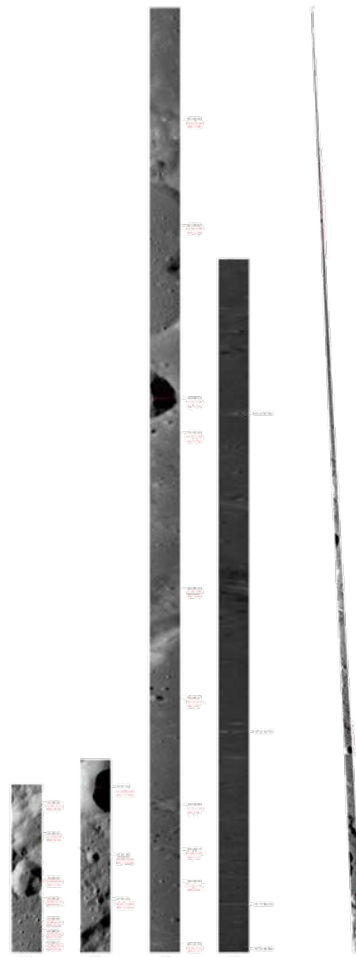




Achievement of CE-1

Photos taken
before Chang'E-1
impact on lunar
surface.

嫦娥一号受控撞月过程CCD影像图



16:01:16
(54.329E, 40.56S)
轨高: 37.7287km

16:01:26
(54.242E, 40.02S)
轨高: 37.1874km

16:01:33
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(54.71E, 47.24S)
轨高: 44.9 km

15:59:15
(54.69E, 47.07S)
轨高: 44.7 km



2. 2nd phase of CLEP

- CE-2 orbiter will be launched in later 2010
- CE-3/4 lunar lander and rover will be launched after 2012. Lunar soft landing and IN-SITU exploration of lunar surface will be conducted.

CE-2 Orbiter is precursor orbiter of soft landing mission.

Mission objective:

key technologies such as LTO launch

X-band TT&C for deep space exploration

acquire high definition image of landing zone for CE-3/4 , improved from 120m to 10m.





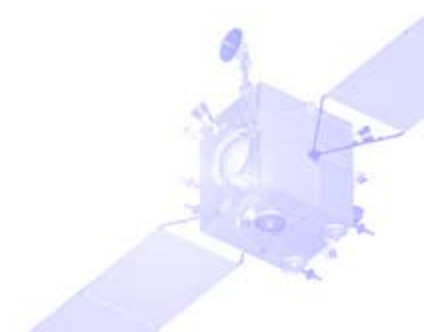
Mission objectives of CE-3/CE-4

CE-3 is a lunar soft landing mission including a lander and a rover.

CE-4 is the backup of CE-3.

Objectives of CE-3/4:

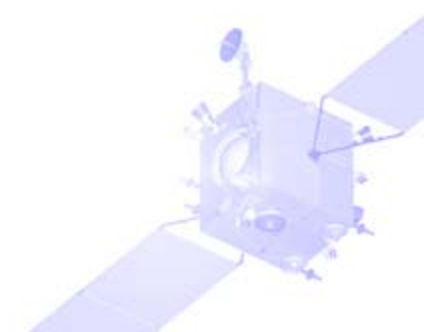
- ⇒ Perform lunar soft landing ,roving, and tele-operation.
- ⇒ Develop lunar soft lander and rover, build up deep space ground antenna, show Chinese capability to execute lunar exploration.
- ⇒ Build up lunar exploration engineering system.





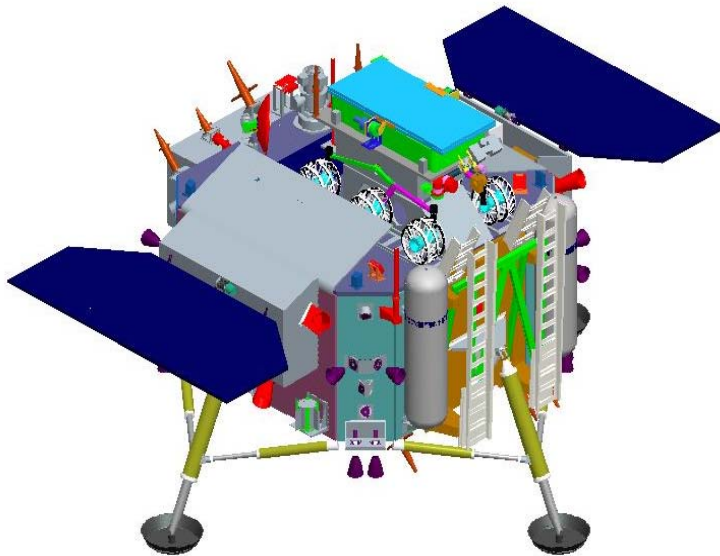
Scientific Objectives of CE-3/CE-4

- ⇒ To study the landscape of lunar surface
- ⇒ To study material composition and usable resources
- ⇒ To study lunar internal structure
- ⇒ To explore the environment among Sun, Earth and Moon and build up telescope based on Moon

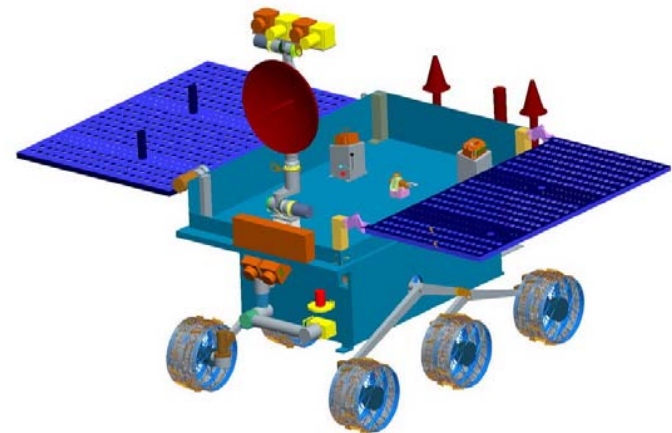




CE-3/CE-4 Lunar Probe



Lunar soft lander



Lunar rover



CE-3/CE-4 Lunar Soft Lander

- ⇒ **11 Subsystems** : GNC, propulsion, power supply, power contribution, TT&C, data handling, structure and mechanism, landing cushion, directional antenna, thermal control and payload.
- ⇒ **Scientific Payloads**: terrain camera, descent imager, telescope and ultra-violet camera.
- ⇒ **Engineering instruments**: monitoring camera, lunar dust detector.





CE-3/4 Lunar rover

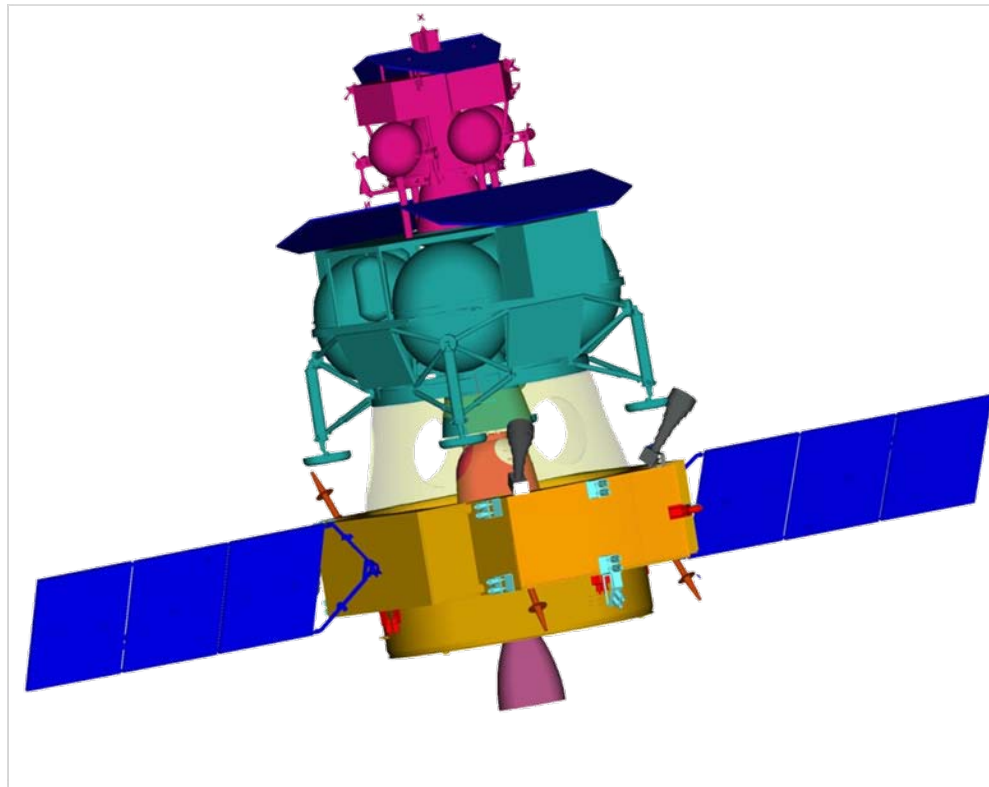
- ⇒ **8 subsystems:** payloads, GNC, thermal control , onboard data handling, power supply, TT&C, structure and mechanics, and locomotion.
- ⇒ **Payloads:** pancamera, lunar subsurface penetration radar, APX and infrared imager spectrometer.





3. 3rd phase of CLEP

⇒ In the third phase of CLEP, a combined probe will perform lunar sample return task.





4. Proposition

⇒ **Information exchange**

Exploration data of Payloads

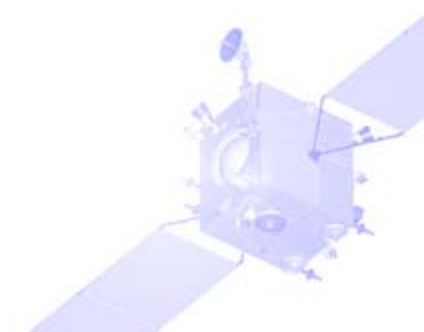
**3-D image of landing zone, such as rainbow,
definition 1m or less**

Lunar dust character

Lunar soil mechanical character

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Academic Exchange Environment





4. Proposition

⇒ **Joint Exploration**

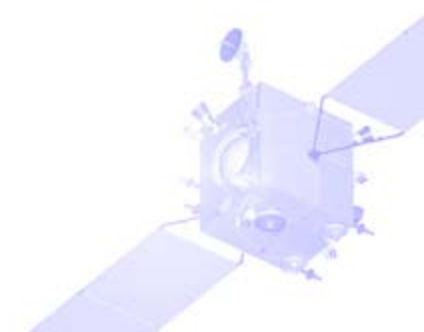
Different zone

Different payloads

Synergy exploration

interferometry, impact wave (moonshake)

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4. Proposition

⇒ **International Lunar Lab**

Openness

Symbol

Union

Limited scale

Permanence

.....

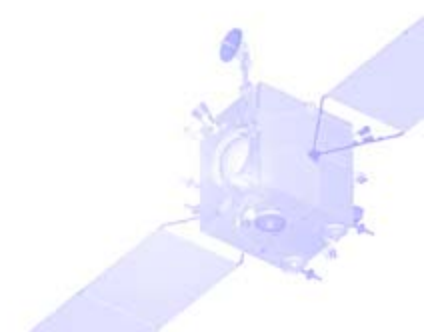
Conception Impetus

propositional companies

conception scheme

folk suggestion from countries

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4. Proposition

Developing research on Lunar Lab supported by 863 plan

- Extensible structure**
- Preparation method on H_2O and O_2**
- Energy system of the Lab**
- Temperature control (underground maybe)**

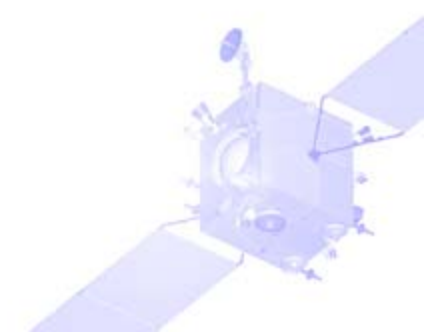
The task will end in 2011.





The end .

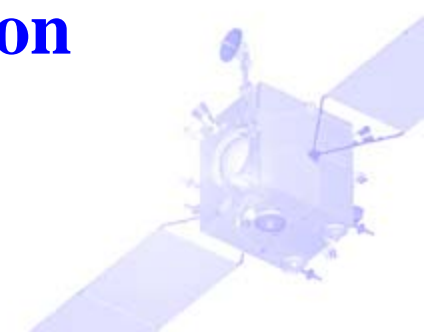
Thanks for your attention!





Contents

- ⇒ **Strategic Plan of CLEP**
- ⇒ **Overview of CE-1**
- ⇒ **Scientific Achievements of CE-1 Orbiter**
- ⇒ **The Latest Updates of CLEP**
 - **CE-2 Orbiter**
 - **CE-3 Lander and Rover**
 - **3rd Phase of CLEP**
 - **Application of CE-1/2 in Mars Exploration**

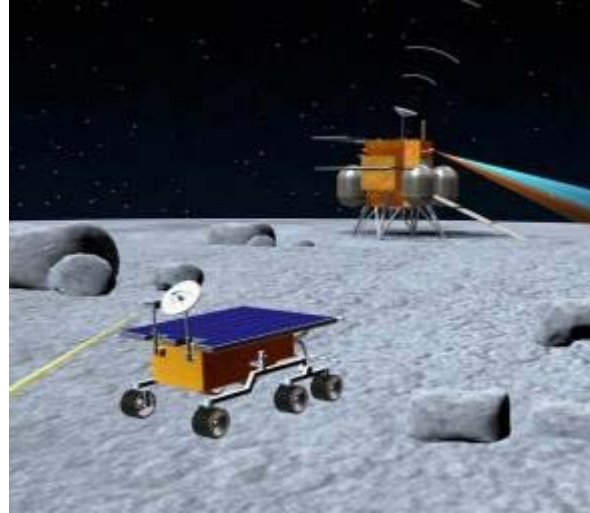




Strategic Plan of CLEP



Orbiting



Landing



Sample and Return

- ⇒ In 1991, Chinese space experts proposed lunar exploration program and conducted some research.
- ⇒ In Nov. 2000, White Paper on China's Space Activities stated to carry out study for deep space exploration focusing on the Moon.
- ⇒ In Jan. 2004, China State Council approved first Phase of China Lunar Exploration Program (CLEP), i.e. Lunar Orbiting Exploration (CE-1) Project.
- ⇒ There are three Phases in CLEP: 2007(orbiting), 2012(landing), and 2016(sample and return).



Development history of CE-1



Engineering System of CE-1 Project

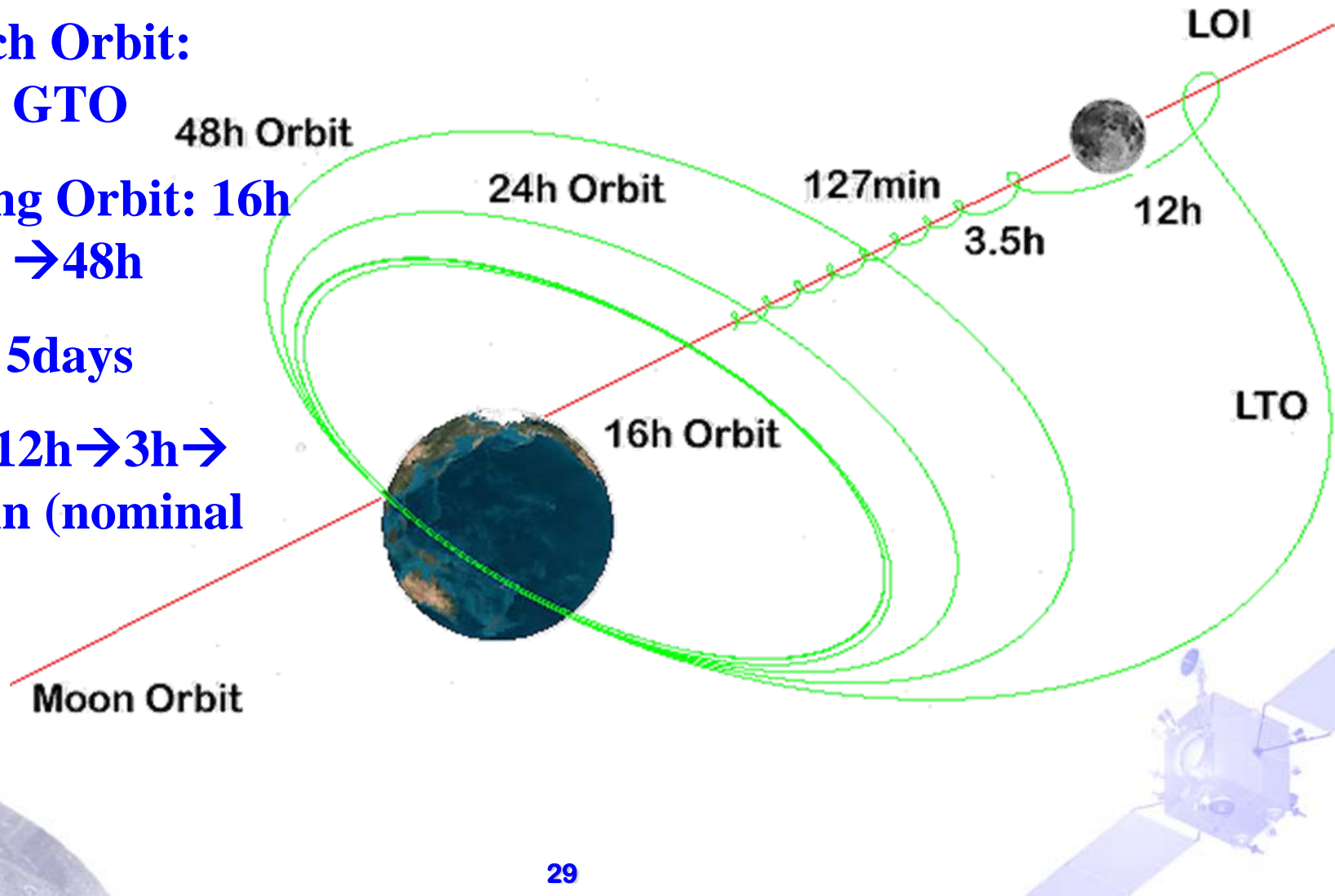
In Feb. 2004, development of Chinese first lunar orbiter CE-1 and Lunar Orbiting Exploration Project were initiated.

CE-1 orbiter was launched in Oct. 24, 2007, who became the third milestone of China space industry.



Flight Profile of CE-1 Orbiter

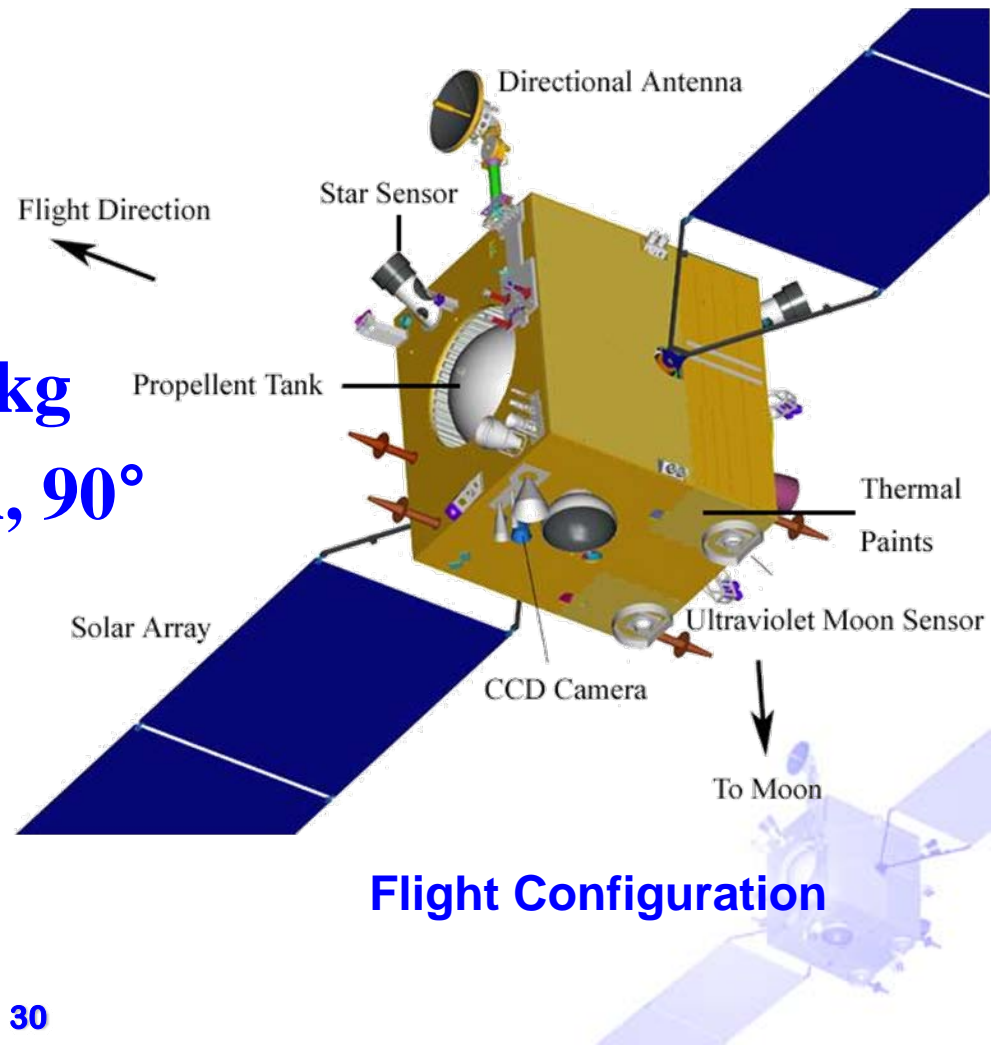
- ⇒ Launch Orbit:
Super GTO
- ⇒ Phasing Orbit: 16h
→ 24h → 48h
- ⇒ LTO: 5days
- ⇒ LOI: 12h → 3h →
127min (nominal
orbit)





Major features of CE-1 Orbiter

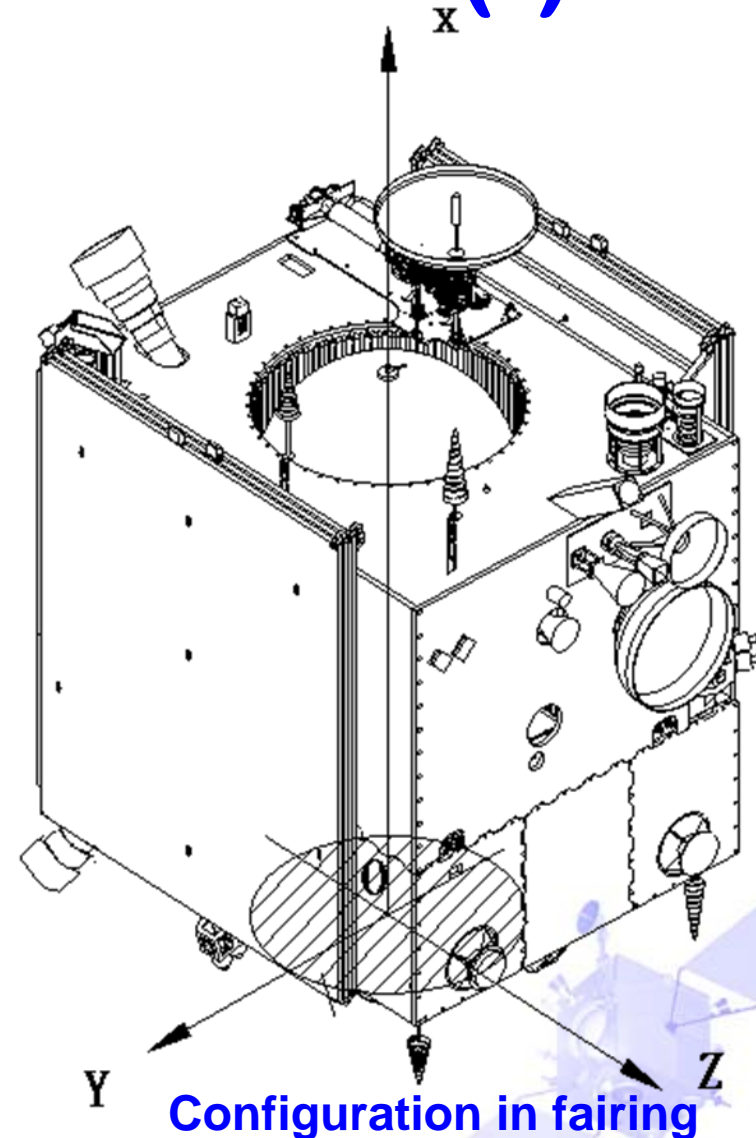
- ⇒ **Launch Mass: 2350kg**
- ⇒ **Dry mass: 1150kg**
- ⇒ **Fuel mass: 1200kg**
- ⇒ **Scientific payloads: 140kg**
- ⇒ **Operation orbit: 200km, 90°**
- ⇒ **Life on orbit: 1 year**





Major Subsystems of CE-1 (I)

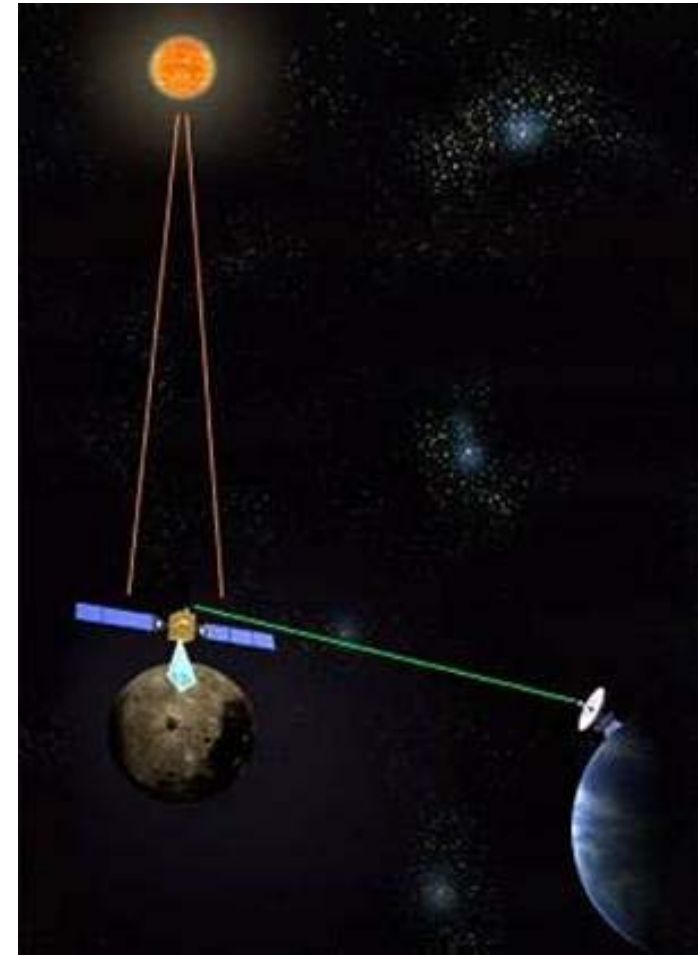
- ⇒ **Structure:** central bearing cylinder, honeycomb panel, box, upper module and bottom module, $2.2\text{m} \times 1.72\text{m} \times 2.2\text{m}$.
- ⇒ **Thermal Control:** active + passive, thermal paint, multilayer thermal blankets and insulation material, heater, sensors, heat pipe and controller.
- ⇒ **Power Supply:** single dimension symmetric solar panel, Si solar cell, 22.7m^2 , Max output 1450W, Ni-H battery (output 48Ah@End Of Life)





Major Subsystems of CE-1 (II)

- ⇒ **GNC:** solar sensor + star tracker + gyroscope + ultra violet sensor, zero moment, reaction wheel with thrusters, three-axis stabilized, pointing accuracy better than $1^\circ(3\sigma)$, stabilization accuracy better than $0.01^\circ/\text{s}$
- ⇒ **Thrusters:** Bi-propellant thrusters with MOH and N_2O_4 , for slow spin, angular rate damper, attitude control and orbit maneuver, $1 \times 490\text{N}$ and $2 \times (6 \times 10\text{N})$ thrusters

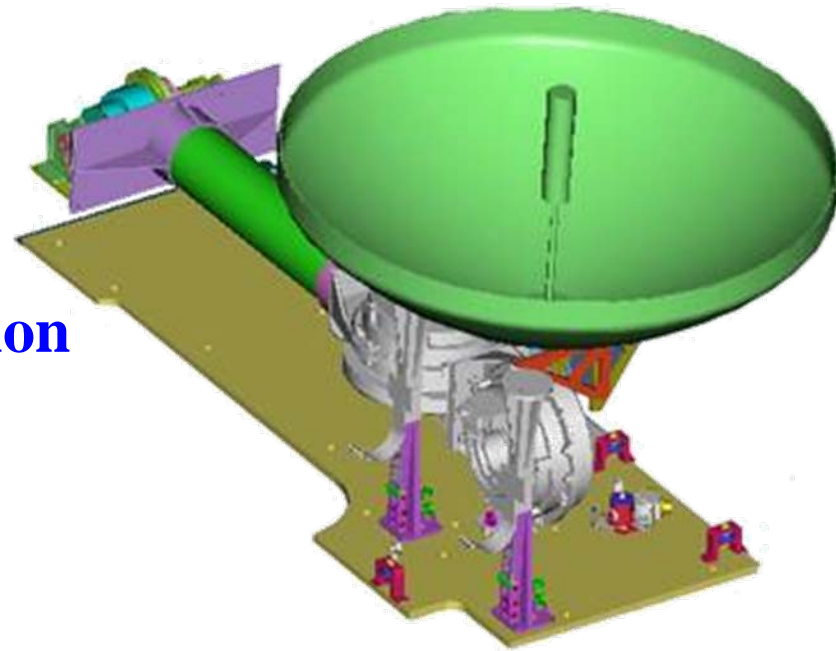


Attitude Control
3 Body Orientation



Major Subsystems of CE-1 (III)

- ⇒ **TT&C:** S-band omni antenna for TT&C, directional antenna for downlink, BPSK, 3Mbps downlink rate, X-band beacon for VLBI
- ⇒ **Directional antenna:** two dimension antenna for semi-sphere coverage, gain not less than +18dBi @ $\pm 5^\circ$
- ⇒ **OBDH:** two-level distributed redundant subsystem, CTU, 4 RTUs, one TCU, one set of redundant SDB.



Directional Antenna



Scientific Payloads of CE-1

- ⇒ **CCD camera/Imager Spectrometer: 3-D images of lunar surface**
- ⇒ **Laser altimeter: surface topology**
- ⇒ **γ -ray/X-ray spectrometer: elements distribution ;**
- ⇒ **Microwave detector: thickness of lunar regolith, He-3;**
- ⇒ **Solar high energy Particle detector/Low energy ion detector: circumstance among Earth, Moon and Sun**



CCD Camera



Laser altimeter



Microwave Detector



Major Events of CE-1

1. 2007-10-24, successfully launched in 0 window
2. 2007-10-31, enter LTO
3. 2007-11-2, en route correction
4. 2007-11-5, first LOI into 12h lunar polar orbit
5. 2007-11-7, enter 127min normal operation orbit
6. 2007-11-20, scientific payloads on
7. 2007-11-26, 1st image issued by Premier Wen Jiabao
8. 2008-2-21, pass first lunar eclipse
9. 2008-8-17, pass second lunar eclipse
10. 2008-10-24, one year after launched, accomplish all scientific and engineering mission goals





Major Events of Extended Mission

1. 2008-12-6, lower orbit into 100km lunar orbit
2. 2008-12-19, lower orbit into 15km×100km orbit
3. 2008-12-20, back into 100km orbit
4. 2009-2-9, pass third lunar eclipse
5. 2009-3-1, impact lunar surface in 52.36°E , 1.5°S by command

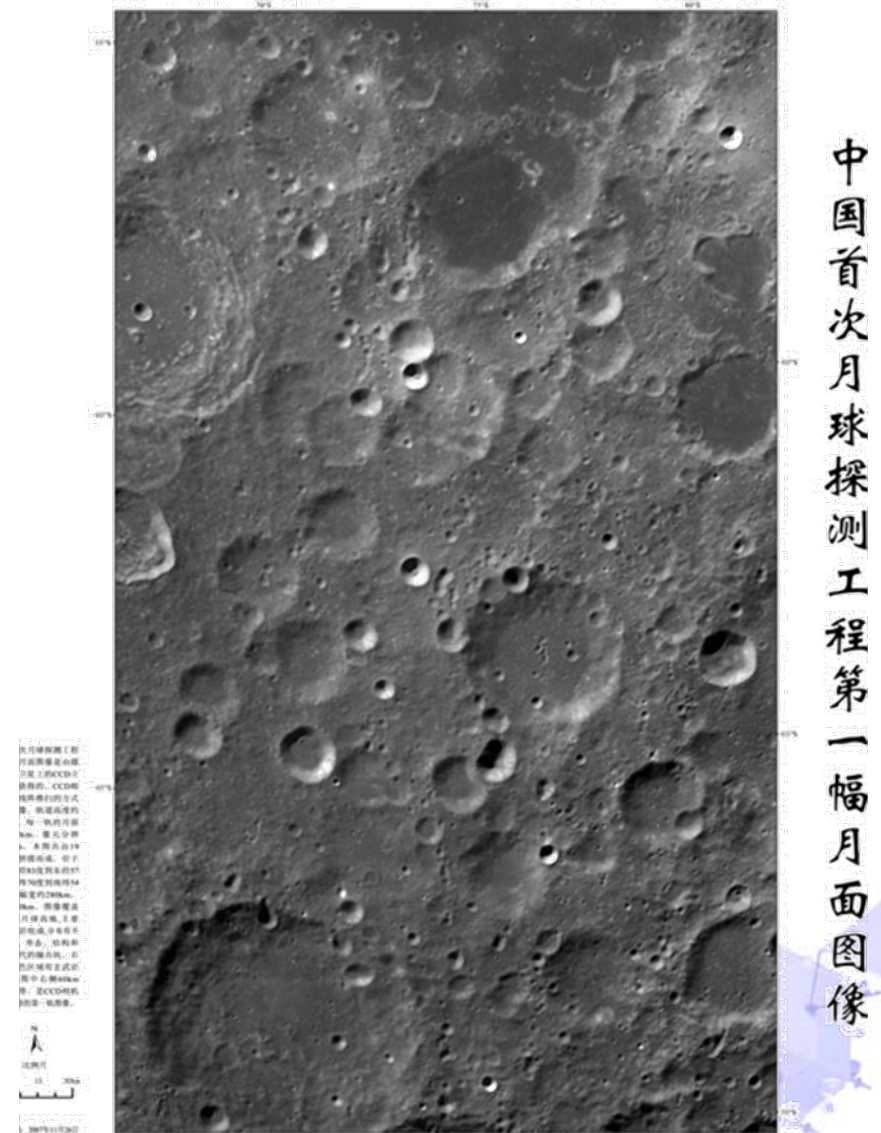
Experiments of CE-1 Extended mission gained more engineering experience for subsequent lunar missions.





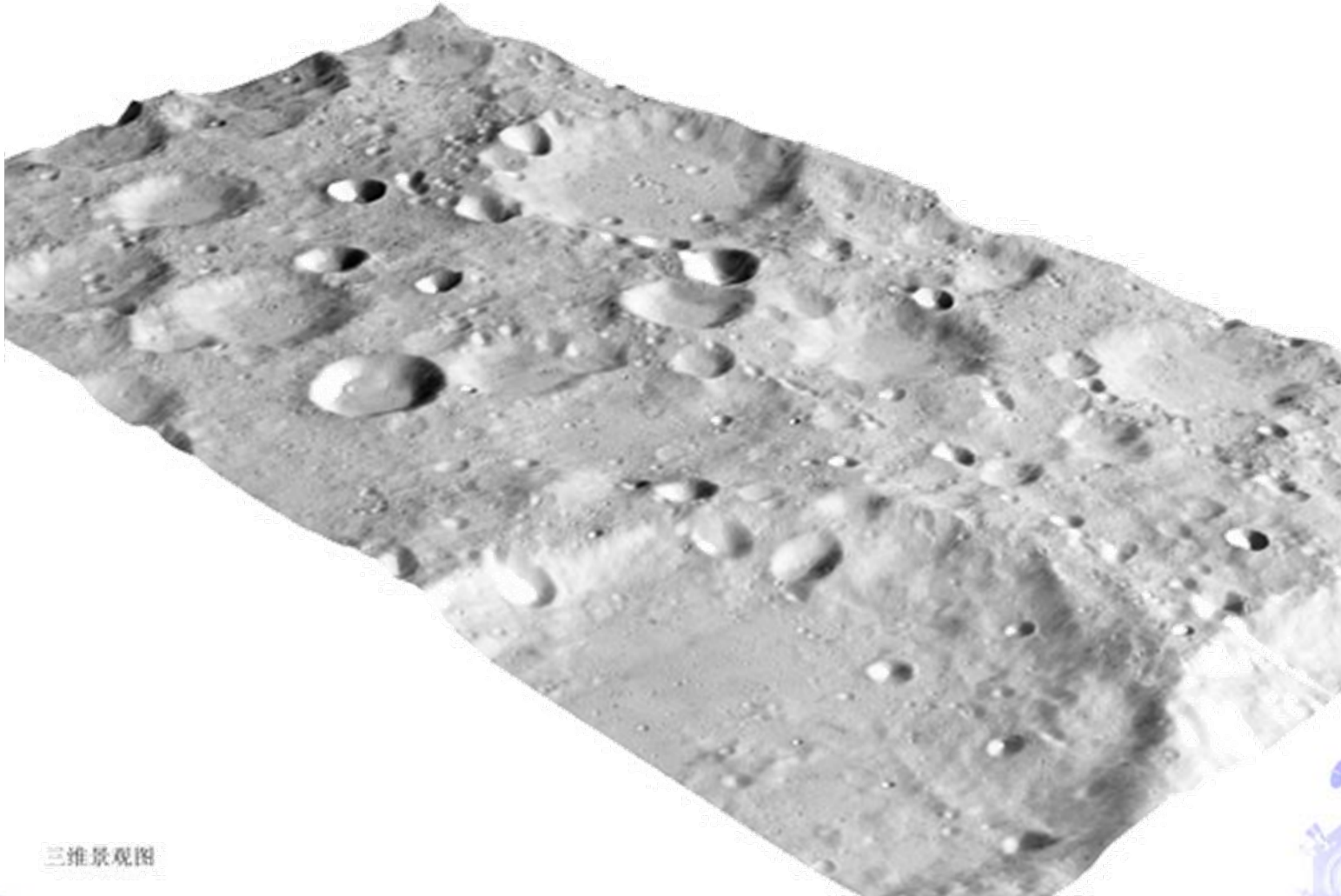
Scientific Achievements of CE-1 (I)

- ⇒ 1st Lunar image by CE-1
- ⇒ Issued in Nov. 26, 2008
- ⇒ Image size: 280km×460km
- ⇒ Pixel resolution: 120m
- ⇒ Location on the Moon:
57°E ~ 83°E , 54°S ~ 70°S





Scientific Achievements of CE-1 (II)



三维景观图

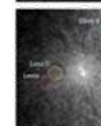
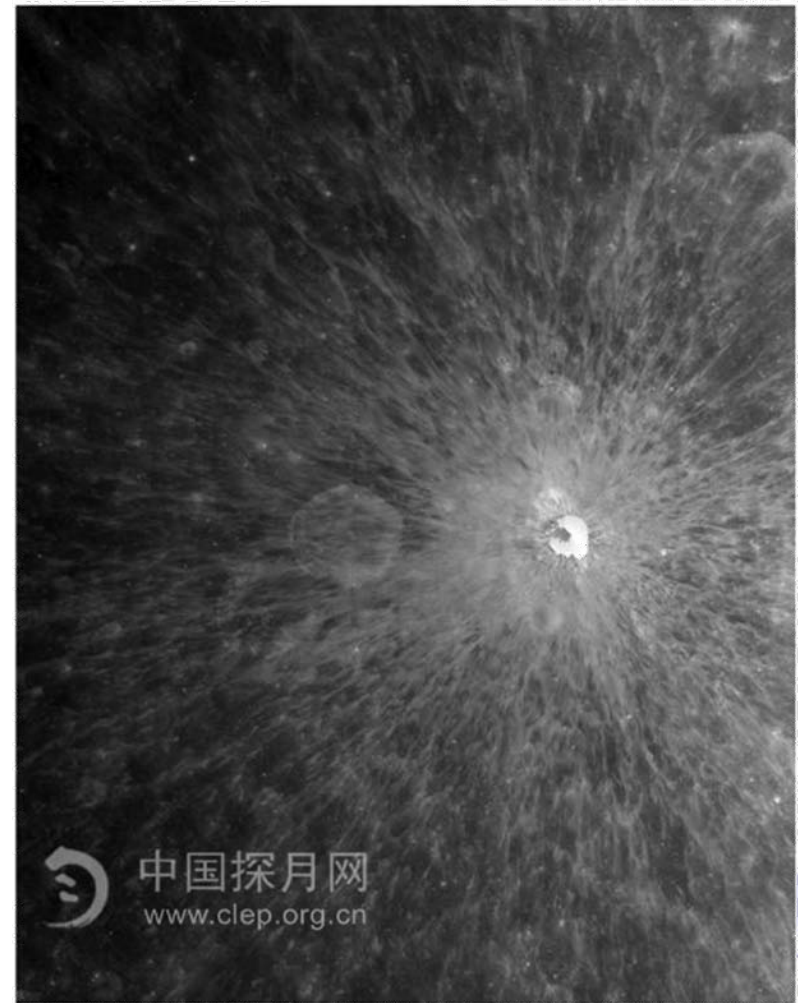
3-D composition of first lunar surface image





Scientific Achievements of CE-1 (III)

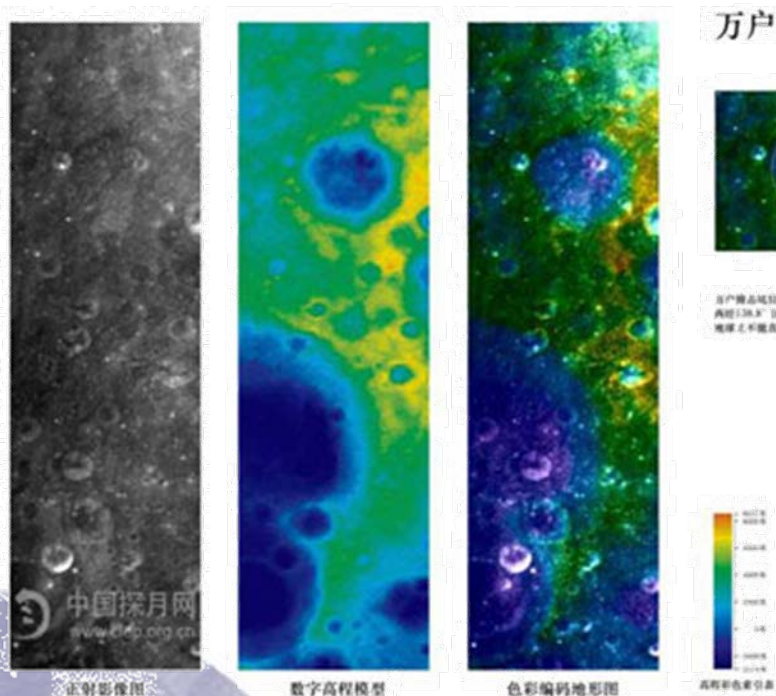
- ⇒ 1st Lunar far side image
- ⇒ Image of crater named by a Chinese Wan Hu who tried making rocket in 1400



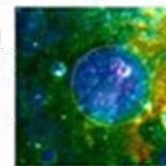
月球背面一个典型的辐射线撞击坑
月球背面Lents撞击坑附近一个典型的辐射线撞击坑，从撞击坑中心放射状抛射大量高亮物质（根据区域地质背景，应该为基性斜长岩碎屑），该类撞击坑形成年代通常较晚。本图是由5轨CCD正视图数据拼接而成的，数据获取时间为2007年12月4日。

月球背面一个典型的辐射线撞击坑

月球背面Lents撞击坑附近一个典型的辐射线撞击坑，从撞击坑中心放射状抛射大量高亮物质（根据区域地质背景，应该为基性斜长岩碎屑），该类撞击坑形成年代通常较晚。本图是由5轨CCD正视图数据拼接而成的，数据获取时间为2007年12月4日。



万户撞击坑



万户撞击坑位于月球背面南纬9.8°、东经138.9°区域，直径52km，是月球上罕见的超新星。



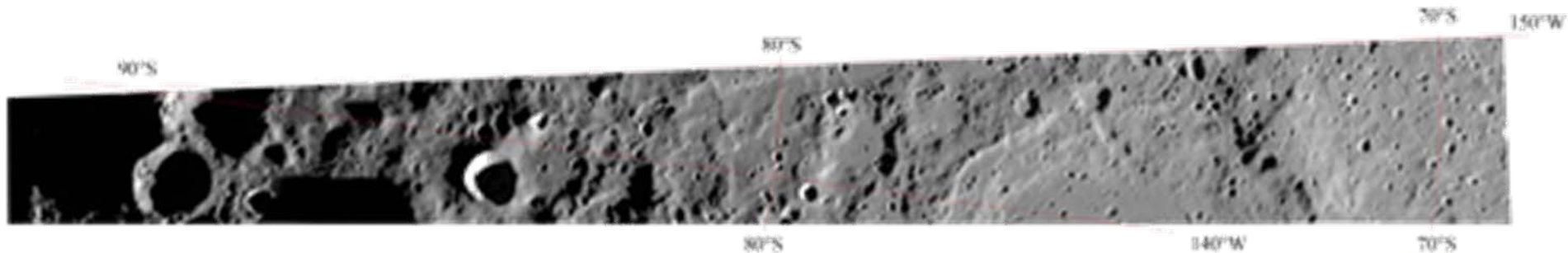
Scientific Achievements of CE-1 (IV)

⇒ Jan. 31, 2008 image of lunar south pole

⇒ Latitude higher than 70°S

⇒ Image quality better than expected

南半球极区影像



月球极区太阳高度角很小，光照很弱，工程设计中原计划只对南北纬70度范围进行光学成像。根据嫦娥1号运行情况和CCD立体相机成像结果分析，在太阳高度角小于设计值15度的高纬度地区，有可能获得清晰的图像，为此进行了极区成像试验。

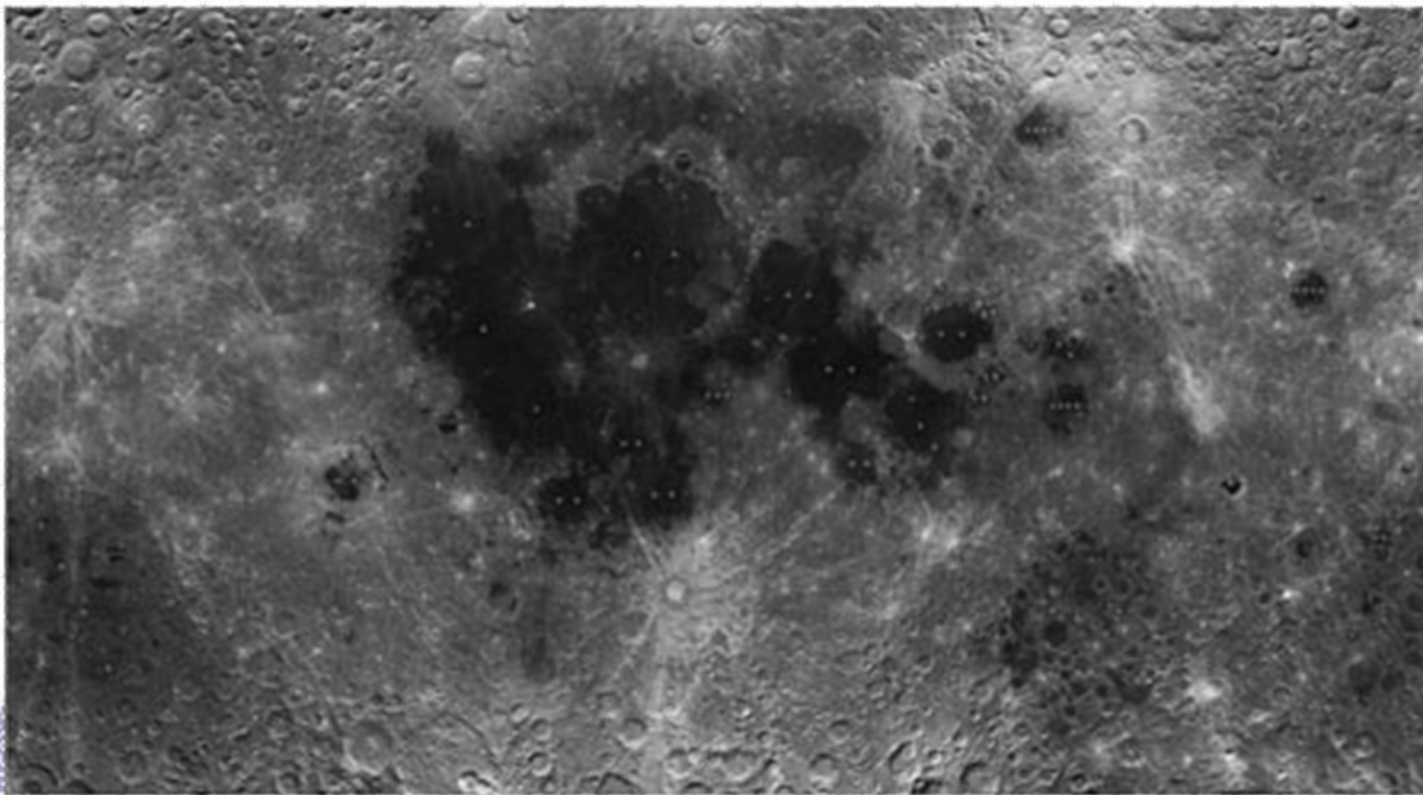
本图是嫦娥一号卫星在2008年1月4日获得的南半球极区影像。由3帧CCD立体相机图像拼接而成（未做几何精校正），位于月球背面西经140度到150度，南纬70度到90度范围。两幅左幅高约60公里，右幅高约80公里，宽约200公里。



Scientific Achievements of CE-1 (V)

⇒ Nov. 12, 2008, lunar global image issued, while first package of scientific data was transferred to scientists

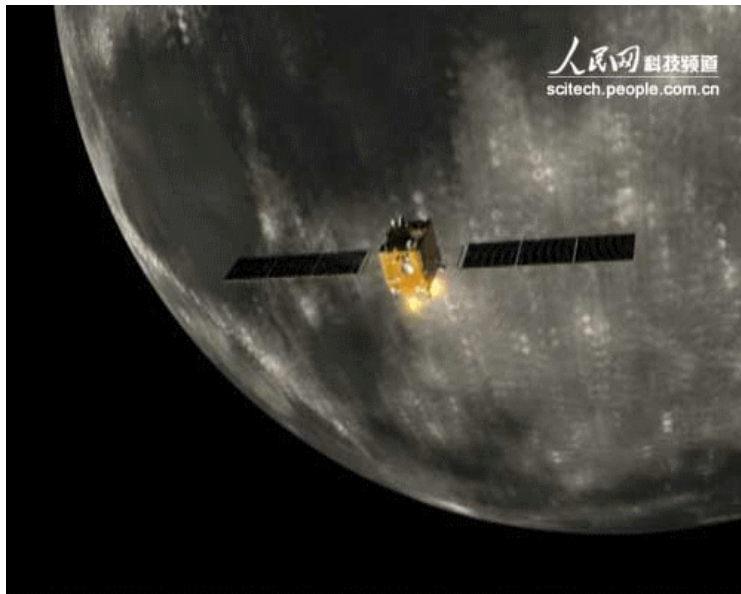
中国首次月球探测工程全月球影像图



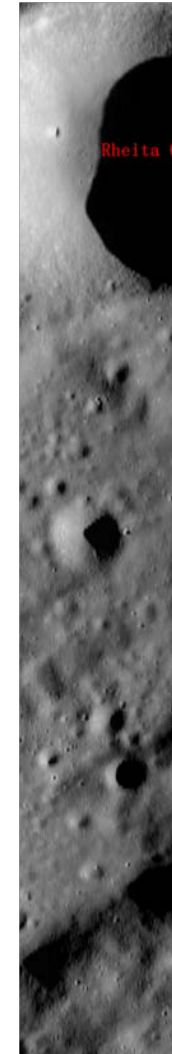
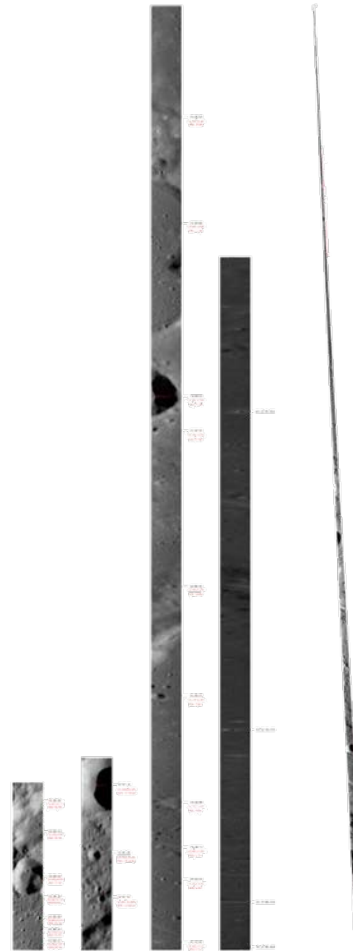


Scientific Achievements of CE-1 (VI)

⇒ Mar. 1, 2009, CE-1 impact the Moon by command



嫦娥一号受控撞月过程CCD影像图



16:01:16
(54.329E, 40.56S)
轨高: 37.7287km

16:01:26
(54.242E, 40.02S)
轨高: 37.1874km

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轨高: 44.9km

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(54.69E, 47.07S)
轨高: 44.7km



The Latest Updates of CLEP

Coming soon after CE-1 success

⇒ CE-2 Orbiter

⇒ CE-3 Lander and Rover

⇒ 3rd Phase of CLEP

⇒ Application of CE-1/2 in Mars Exploration





CE-2 Orbiter

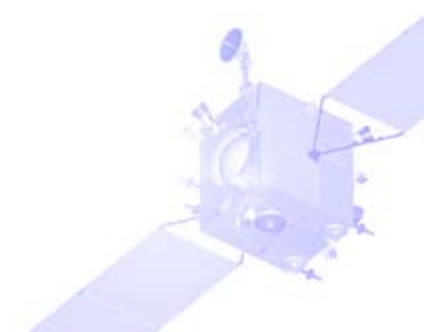


- ⇒ Once was a backup of CE-1
- ⇒ CE-2 will be launched in Oct. 2010
- ⇒ Different payloads and orbit from CE-1
- ⇒ Demonstrate technologies for subsequent deep space exploration



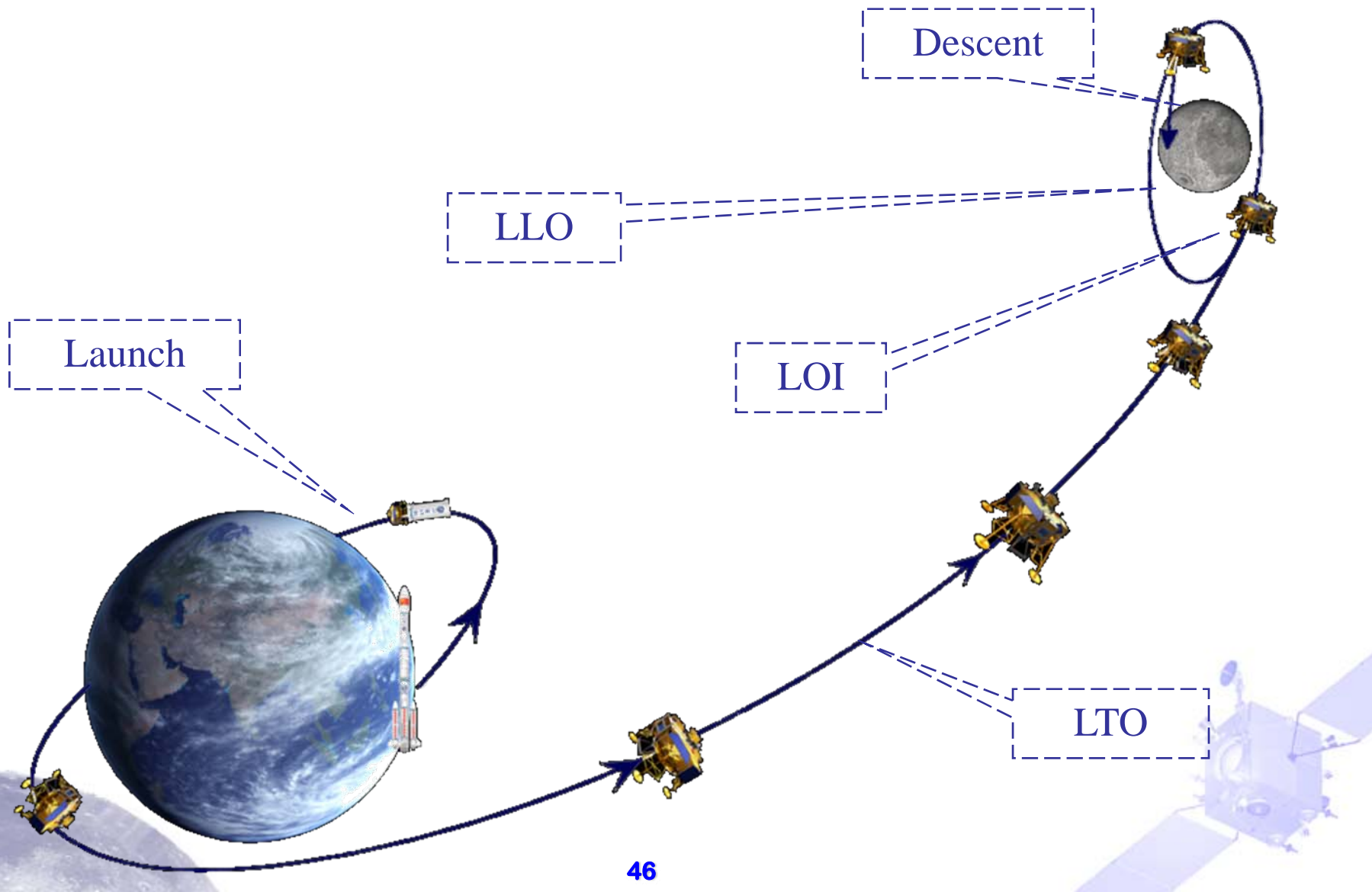
Scientific Objectives of CE-3

- ⇒ To investigate the landscape of lunar surface and geological structure of Moon
- ⇒ To investigate material composition and usable resources of lunar surface
- ⇒ To study lunar internal structure
- ⇒ To build up an observatory based on Moon



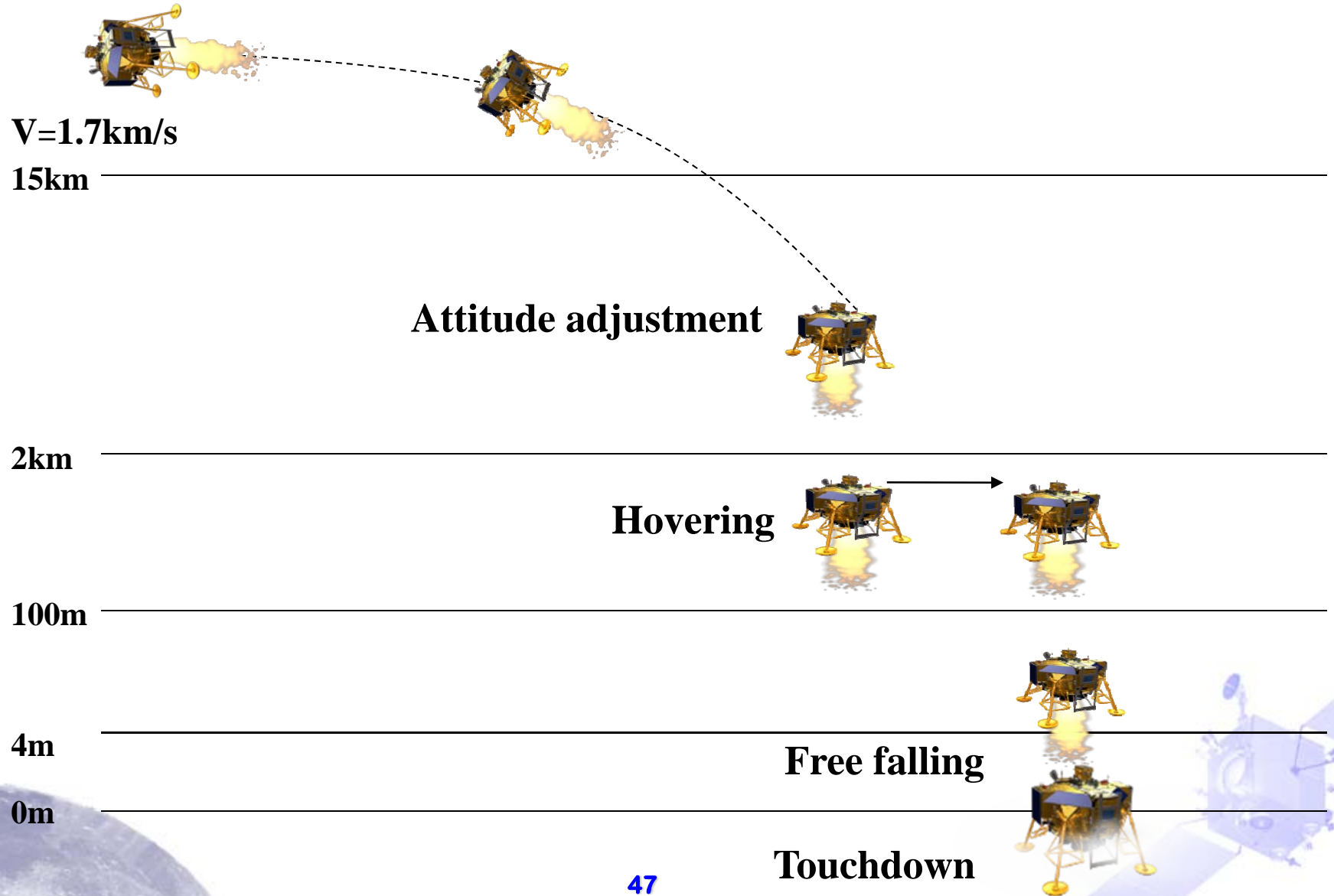


Flight Profile Of CE-3





Power Descent of CE-3 Lander





Engineering System of CE-3 mission



Lander + Rover

64m/35m/50m + VLBI



CZ-3B , Xichang



Developments of Lander & Rover



Structure model



Wheels of the rover prototype



Impact test of landing leg

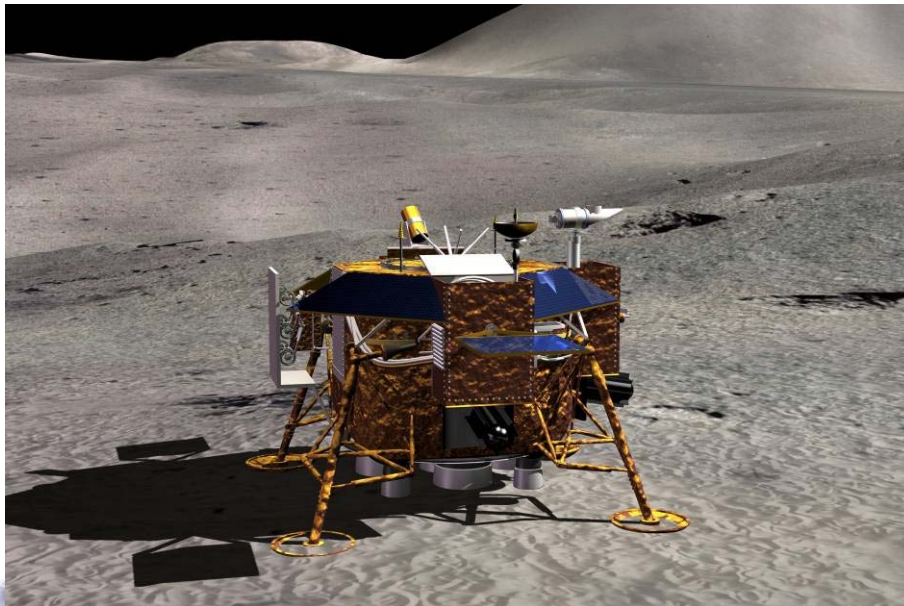


Rover prototype under test

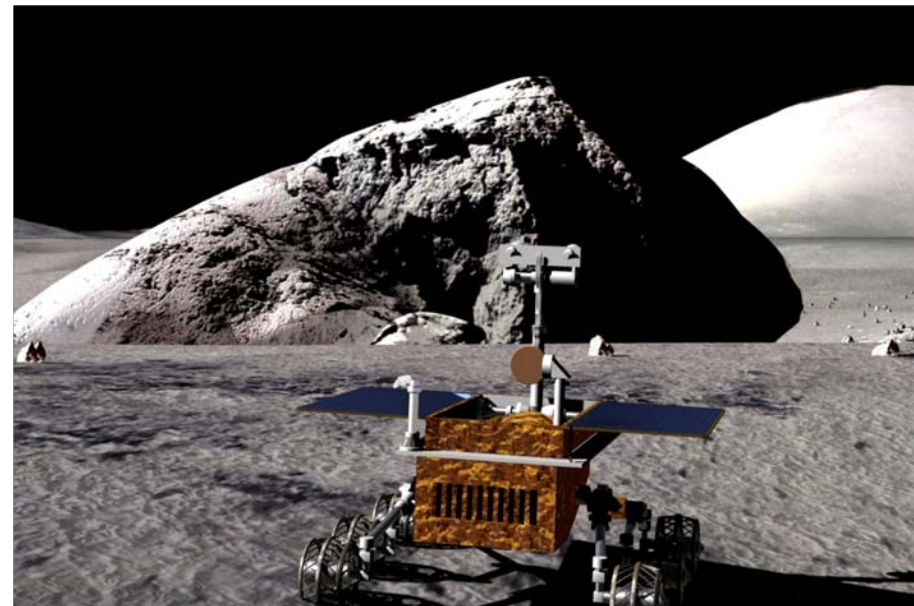


Overview of CE-3 Lander & Rover

- ⇒ Launch date in schedule: later in 2012
- ⇒ Launch mass: 3750kg
- ⇒ Landing site: Rainbow (near side)



Lander



Rover



3rd Phase of CLEP

- ⇒ **Ascending stage, return capsule, orbiter**
- ⇒ **Launch rockets: CZ-5E**
- ⇒ **Robotic sample and return**
- ⇒ **Performed in 2016**
- ⇒ **Key technologies for sample and return**
- ⇒ **Study lunar sample in ground lab**
- ⇒ **Learn technologies necessary for robotic exploration and future more deeper space exploration even manned lunar exploration**





The End

Thanks for your attention

